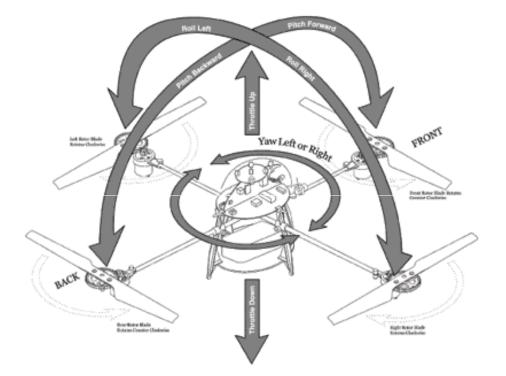


# **REAL TIME EMBEDDED CONTROL SYSTEM**



#### Application: VTOL Aircrafts – Quad-Rotor Helicopter

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## A few acronyms

• UAV

**Unmanned Aerial Vehicles** 

• VTOL

Vertical Take-Off and Landing



### **Challenges:**

**Complex laws of Aerodynamics** 

No room for error!

#### **Goal / Objective**

To develop a platform to stabilize flying vehicles which are able to make Vertical Take-Offs and Landings



#### The Embedded System:-

**RABBIT RCM3400 MCU** 

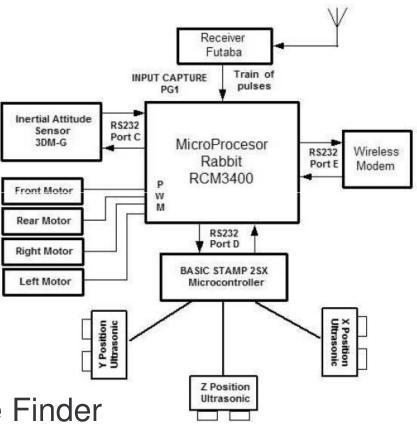
512Kb FLASH

- 512Kb SRAM
- 6 Serial Ports (A E)
- 4 PWM outputs

**SENSORS**:

IMU 3DM-G Attitude Sensor

3 x SRF04 Ultrasonic Range Finder (data gathered by a BS2X Basic Stamp MCU)





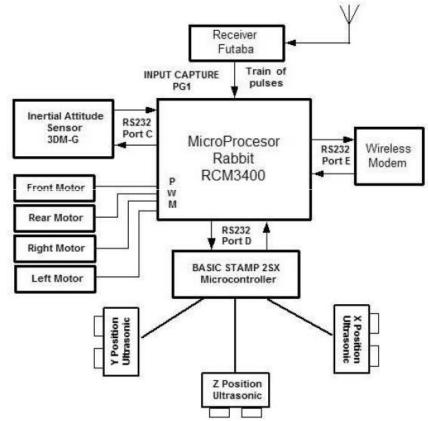
# The Embedded System: ACTUATORS DC Motors driven by PWM Inertial A Sens SOMMUNICATIONS

RS-232

Attitude Sensor Wireless Modem

**FUTABA** Receiver

Wireless Modem





#### **Software Development Tools**

IDE: "Dynamic C"

Multi tasking – Frequently referred to as *"virtual parallel processing"* 

Takes advantage of natural delays in each task to increase overall system performance



#### **Software Architecture**

**Begin** – Initialization

Run Scheduler as follows:

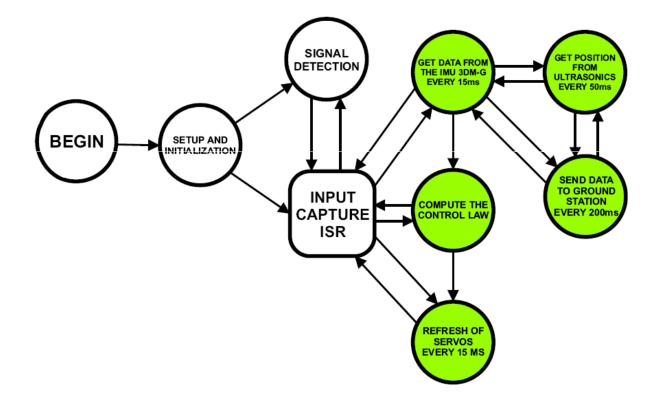
- Get attitude data from IMU every 15ms
- The IMU gets position info from the ultrasonics every 50ms
- Send data to ground station every 200ms
- Refresh Servos every 15ms

Input Capture ISR

- Triggered by falling edge of incoming pulses
- Detect received pulses and check validity of data
- Calculate the control law



#### **Software Architecture**





### The Input Capture ISR

If a falling edge is detected and the signal is valid, the periodical train of 7 data pulses must exist:-

Pulse	Name	Description
0	Flank	Used for ISR synchronization
1	Yaw	Yaw set point
2	Pitch	Offset adjustment for pitch
3	Power	Value of the thrust of the motors
4	Roll	Offset adjustment for roll
5	Auto-Manual	Switch for Auto-manual mode
6	Yaw-Trim	Yaw Fine adjustment



#### **Conclusion and further work**

Designed an embedded control system for VTOL aircrafts

System was tested on a quad control helicopter

System tests proved Non-Linear control was more robust than the conventional linear

Further work includes use of a GPS sensor to measure position of the aircraft



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